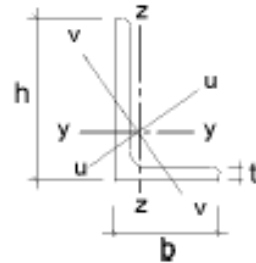
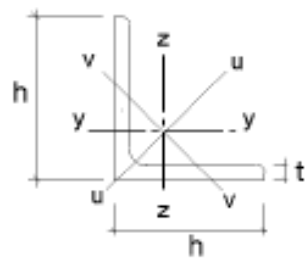
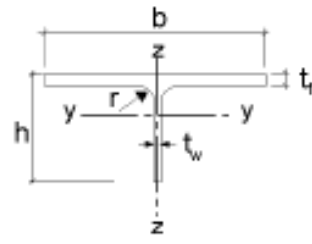
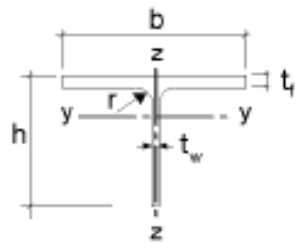
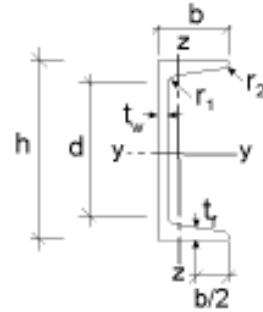
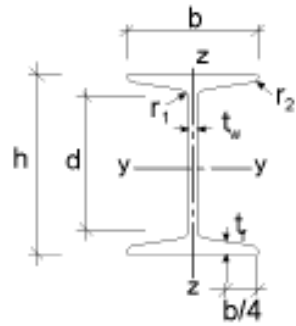
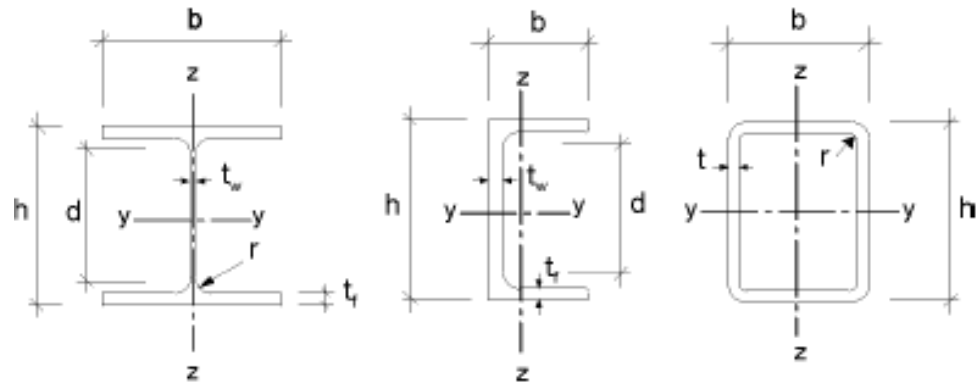


Stability of Industrial Columns

According to EN 1993-1: 2005
and the Italian Code

Symbols



Buckling resistance

6.3 Buckling resistance of members

6.3.1 Uniform members in compression

6.3.1.1 Buckling resistance

(1) A compression member should be verified against buckling as follows:

$$\frac{N_{Ed}}{N_{b,Rd}} \leq 1,0$$

where N_{Ed} is the design value of the compression force;

$N_{b,Rd}$ is the design buckling resistance of the compression member.

(3) The design buckling resistance of a compression member should be taken as:

$$N_{b,Rd} = \frac{\chi A f_y}{\gamma_{M1}} \quad \text{for Class 1, 2 and 3 cross-sections} \quad (6.47)$$

$$N_{b,Rd} = \frac{\chi A_{eff} f_y}{\gamma_{M1}} \quad \text{for Class 4 cross-sections} \quad (6.48)$$

where χ is the reduction factor for the relevant buckling mode.

NOTE For determining the buckling resistance of members with tapered sections along the member or for non-uniform distribution of the compression force second order analysis according to 5.3.4(2) may be performed. For out-of-plane buckling see also 6.3.4.

Buckling curves

(1) For axial compression in members the value of χ for the appropriate non-dimensional slenderness $\bar{\lambda}$ should be determined from the relevant buckling curve according to:

$$\chi = \frac{1}{\Phi + \sqrt{\Phi^2 - \bar{\lambda}^2}} \quad \text{but } \chi \leq 1,0 \quad (6.49)$$

where $\Phi = 0,5 \left[1 + \alpha(\bar{\lambda} - 0,2) + \bar{\lambda}^2 \right]$

$$\bar{\lambda} = \sqrt{\frac{Af_y}{N_{cr}}} \quad \text{for Class 1, 2 and 3 cross-sections}$$

$$\bar{\lambda} = \sqrt{\frac{A_{eff} f_y}{N_{cr}}} \quad \text{for Class 4 cross-sections}$$

α is an imperfection factor

N_{cr} is the elastic critical force for the relevant buckling mode based on the gross cross sectional properties.

(2) The imperfection factor α corresponding to the appropriate buckling curve should be obtained from Table 6.1 and Table 6.2.

Table 6.1: Imperfection factors for buckling curves

Buckling curve	a_0	a	b	c	d
Imperfection factor α	0,13	0,21	0,34	0,49	0,76

Slenderness definition

- (1) The non-dimensional slenderness $\bar{\lambda}$ is given by:

$$\bar{\lambda} = \sqrt{\frac{Af_y}{N_{cr}}} = \frac{L_{cr}}{i} \frac{1}{\lambda_1} \quad \text{for Class 1, 2 and 3 cross-sections} \quad (6.50)$$

$$\bar{\lambda} = \sqrt{\frac{A_{eff}f_y}{N_{cr}}} = \frac{L_{cr}}{i} \sqrt{\frac{A_{eff}}{A}} \frac{1}{\lambda_1} \quad \text{for Class 4 cross-sections} \quad (6.51)$$

where L_{cr} is the buckling length in the buckling plane considered

i is the radius of gyration about the relevant axis, determined using the properties of the gross cross-section

$$\lambda_1 = \pi \sqrt{\frac{E}{f_y}} = 93,9\varepsilon$$

$$\varepsilon = \sqrt{\frac{235}{f_y}} \quad (f_y \text{ in N/mm}^2)$$

NOTE B For elastic buckling of components of building structures see Annex BB.

Buckling curves

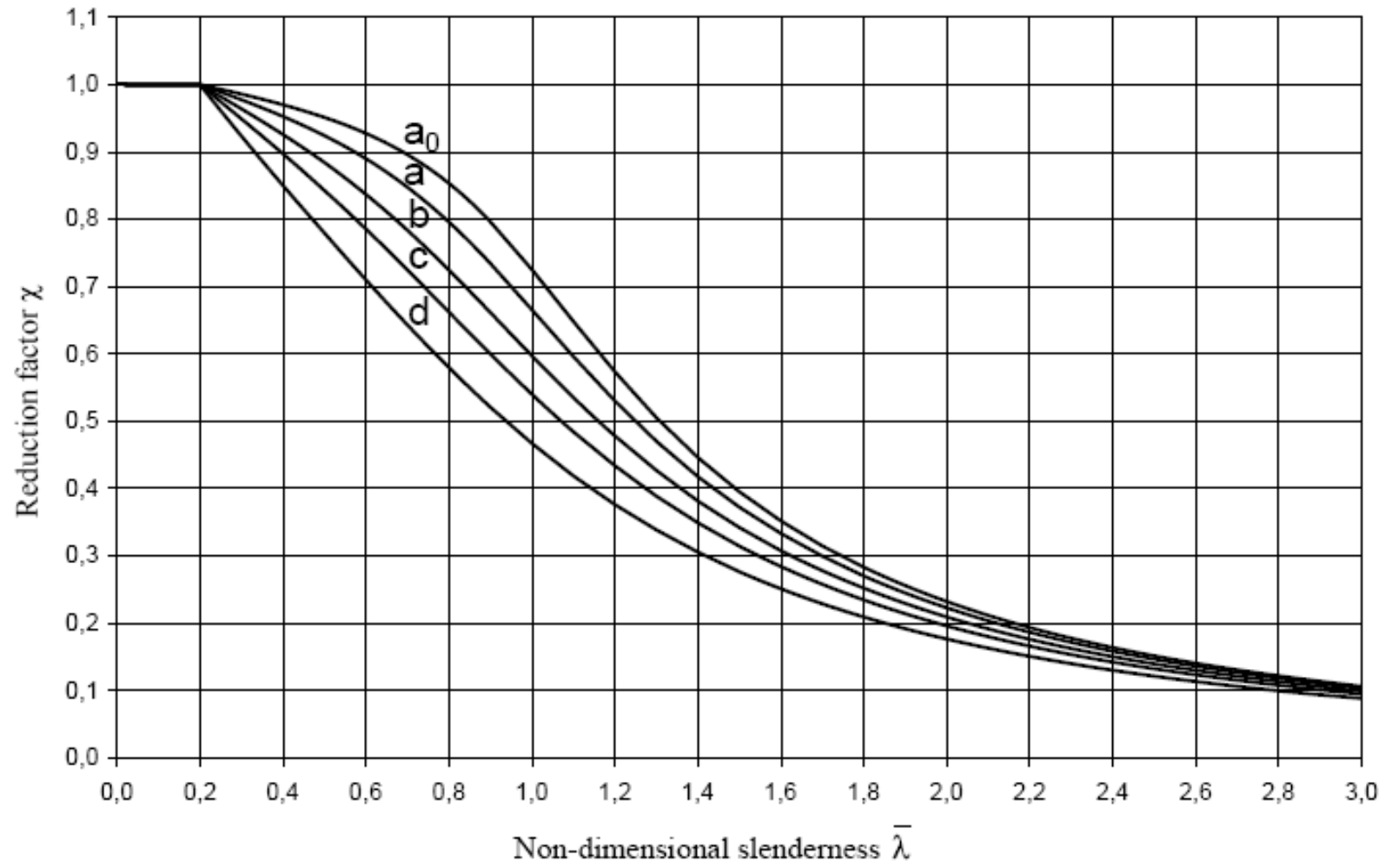
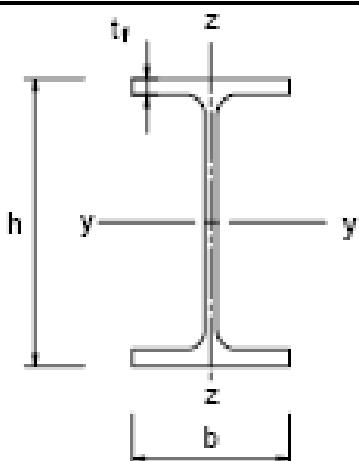
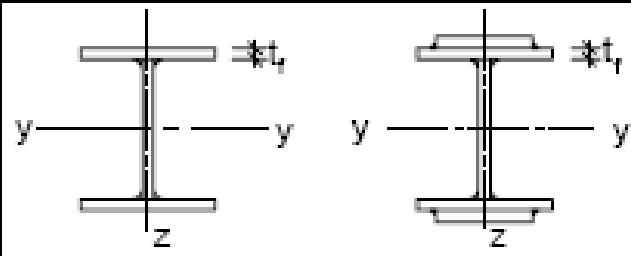
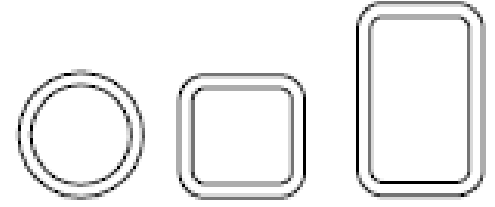
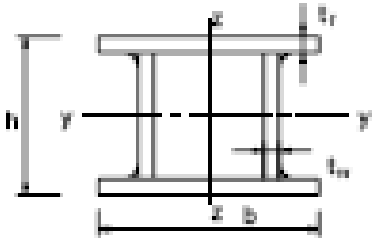
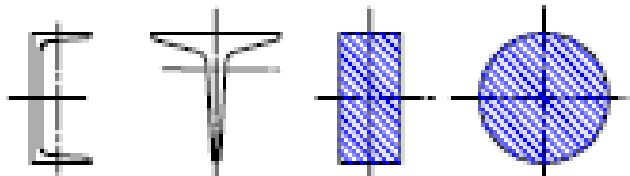
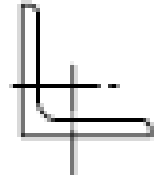


Figure 6.4: Buckling curves

Buckling curves - Sections

Cross section		Limits	Buckling about axis	Buckling curve		
				S 235 S 275 S 355 S 420	S 460	
Rolled sections		$h/b > 1,2$	$t_f \leq 40 \text{ mm}$ $40 \text{ mm} < t_f \leq 100$	y-y z-z	a a ₀	
				y-y z-z	b a	
		$h/b \leq 1,2$	$t_f \leq 100 \text{ mm}$	y-y z-z	b c	a a
			$t_f > 100 \text{ mm}$	y-y z-z	d d	c c
Welded I-sections		$t_f \leq 40 \text{ mm}$	y-y z-z	b c	b c	
		$t_f > 40 \text{ mm}$	y-y z-z	c d	c d	
Hollow sections		hot finished	any	a	a ₀	
		cold formed	any	c	c	

Buckling curves - Sections

Cross section	Limits	Buckling about axis	Buckling curve	
			§ 235 § 275 § 355 § 420	§ 460
	generally (except as below)	any	b	b
	thick welds: $a > 0,5t_f$ $b/t_f < 30$ $h/t_w < 30$	any	c	c
		any	c	c
		any	b	b

Stability of beam-columns

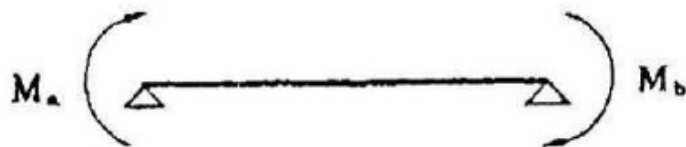
$$\frac{N_{Ed} \cdot \gamma_{M1}}{\chi_{\min} \cdot f_{yk} \cdot A} + \frac{M_{yeq,Ed} \cdot \gamma_{M1}}{f_{yk} \cdot W_y \cdot \left(1 - \frac{N_{Ed}}{N_{cr,y}}\right)} + \frac{M_{zeq,Ed} \cdot \gamma_{M1}}{f_{yk} \cdot W_z \cdot \left(1 - \frac{N_{Ed}}{N_{cr,z}}\right)} \leq 1$$

Definition of the equivalent bending moment

Linear variation of the bending moment throughout the beam-column axis

$$|M_a| \geq |M_b|$$

$$M_{eq,Ed} = 0,6 \cdot M_a - 0,4 \cdot M_b \geq 0,4 \cdot M_a$$



Non-linear variation of the bending moment throughout the beam-column axis

Average value of the bending moment throughout the axis

$$M_{eq,Ed} = 1,3 \cdot M_{m,Ed}$$

$$0,75 \cdot M_{\max,Ed} \leq M_{eq,Ed} \leq M_{\max,Ed}$$